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Hydrogen sensing in Titan's atmosphere: Motivations and techniques

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19 Abstract

20 We summarize the observations and context for molecular hydrogen (H₂) in Titan's atmosphere where it 21 is the third most abundant gas. Hydrogen escapes to space but is replenished by methane photochemistry. An open question is whether there are sources and/or sinks in the surface and subsurface : sources might 22 23 include serpentinization reactions in the deep interior while sinks might involve reactions with acetylene 24 mediated by chemical or even biological catalysts. Cassini data provide weak evidence of a surface sink, and also point to variations with latitude of the tropospheric hydrogen abundance, so further 25 26 measurements would be of value. We demonstrate that a simple solid-state sensor can provide the 27 required measurement precision in an oxygen-free atmosphere, and consider how measurements on a 28 mobile platform may inform the question of sources and sinks. We underscore the importance of 29 simultaneous methane and hydrogen measurements: whereas the stratosphere is a hydrogen source and methane sink, serpentinization could be a subsurface source of both gasses. 30

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34 Introduction

35 Hydrogen is the third-most abundant (~0.1%) gas in Titan's atmosphere, after nitrogen (~95%) and

36 methane (~5%). Although some early detections were reported (e.g. Trafton (1972)) which seem to have

been spurious, the abundance was well-determined as 0.002 +/- 0.001 in Voyager infrared spectra by

38 Samuelson et al. (1982). Analysis of the same data was refined by Courtin et al. (1995) to 0.001 +/-

39 0.0004.

This abundance reflects the balance between production and escape. Methane photolysis by ultraviolet
light leads to the production of heavier organic compounds, and hydrogen. Simplistically, the light
hydrogen molecules can 'easily' escape Titan's gravity (e.g. Hunten, 1973), making the photolysis process
largely irreversible, although the details of the transport and loss processes are complex (e.g. Lebonnois et
al., 2003; Strobel, 2010).

Even though its abundance is small, Titan's hydrogen plays a meaningful role in Titan's climate, in that
N₂-H₂ absorption fills what would otherwise be a thermal window in the atmosphere (Mckay et al., 1990,
1991). Hydrogen is therefore a significant greenhouse gas that is produced by photochemistry of the main
greenhouse gas (in this respect, hydrogen on Titan is analogous to ozone on Earth).

The Cassini mission yielded estimates of the hydrogen abundance at high altitudes (~1000km) via direct in-situ measurements with the Ion and Neutral Mass Spectrometer (INMS, e.g. Cui et al. (2008)). Additional measurements were made by the Huygens probe throughout its descent (~140km down to the surface at 10°S) with the Gas Chromatograph Mass Spectrometer (GCMS, Niemann et al., 2010). The molecular hydrogen mole fraction was $(1.01 \pm 0.16) \times 10^{-3}$ in the atmosphere and (9.90 ± 0.17) $\times 10^{-3}$ on the surface.

56 More recently, Courtin et al. (2012) found in spectra from the Cassini Composite Infrared Spectrometer 57 (CIRS) that the latitudinal distribution of H_2 in Titan's troposphere appears to be non-uniform, with a mole fraction above 50°N larger than the globally-averaged value. This is unexpected since the lifetime of 58 hydrogen is expected to be long, and Courtin et al. (2012) speculate that the northern hemisphere was 59 60 enriched due to downwelling of hydrogen-rich stratospheric air in that winter season (the observations were acquired in 2006-2007). Courtin et al. (2012) note that the implied gradients are "not consistent with 61 the present understanding of dynamics and chemistry in Titan's atmosphere". It seems likely that further 62 insights may accrue from analysis of later Cassini observations. 63 The vertical profile of hydrogen has attracted some interest : Strobel (2010) noted that a model of 64 photochemical production and escape could not simultaneously match the thermospheric (INMS) 65 66 abundance and the stratospheric/tropospheric abundance (CIRS/GCMS) without also invoking a surface sink. Although this may be a model-dependent result, such a surface sink is at least superficially 67 68 consistent with the small decline in the GCMS H₂ abundance towards the surface, and in the following section we consider possible surface sources and sinks. The measurements and processes pertaining to 69 70 hydrogen are summarized in schematic form in figure 1.



Figure 1. Summary of the observations of hydrogen abundance and the processes influencing it on Titan.

75 Surface Sources and Sinks for Hydrogen

76 The photochemical source for molecular hydrogen in the atmosphere is generally established overall as 77 methane photolysis (although both production of H₂ molecules and removal of atomic hydrogen may 78 occur in the process of stratospheric haze formation -e.g. Sekine et al., 2008). There is no need for a 79 subsurface source to explain the observed hydrogen abundance (and indeed, it seems any such source 80 must be overpowered by a stronger sink) but the observation of molecular hydrogen in the plume of Enceladus (Waite et al., 2017) suggests by analogy that Titan might nonetheless have a similar such 81 82 source. Hydrogen was observed in the Enceladus plume at somewhat higher abundances (0.4-1.4%) than 83 in Titan's atmosphere, and is presumed to have been formed by hydrothermal reactions in Enceladus' warm interior. The serpentinization reaction specifically can yield hydrogen and methane from the 84 85 reaction of silicates with water and carbon dioxide, and seems consistent (Waite et al., 2017) with the methane abundance in the Enceladus plume (0.1-0.3, a H₂:CH₄ ratio of 1-14). Should such similar 86 87 reactions occur on Titan - and the presumed long-term geological source of methane on Titan has not 88 been identified – a surface hydrogen vent might be expected to also be a methane source. Serpentinization has been suggested as a possibly important process in providing available free energy for biological 89 90 systems on the early Earth or Mars (e.g. Schulte et al., 2006).

As for sinks, note that the solubility of hydrogen in liquid methane and ethane is very low, and thus Titan's seas do not make an appreciable reservoir. Dobouloz et al. (1989) calculated saturation mixing ratios of hydrogen in Titan's seas as of the order of 1 ppm : roughly speaking, then, there is about 10 times less hydrogen in a cubic meter of sea liquid as there is in a cubic meter of air above it. A physical absorption onto solids (notably the haze) seems similarly to be unviable as a long-term sink.

96 Chemical removal of hydrogen is the most plausible sink possibility. Among a number of possible 97 reactions, perhaps the most likely is the hydrogenation of acetylene. This reaction has the advantage that 98 acetylene is an abundant product of photochemistry, and the reaction is exothermic, as pointed out by 99 McKay and Smith (2006). However, the reaction is kinetically inhibited at Titan's low temperatures. 100 In fact, the hydrogenation of acetylene is a well-studied reaction on Earth in industrial settings (e.g. Bos 101 and Westerterp, 1993). Specifically, the production of the plastic polyethylene from ethylene relies on 102 catalysts for polymerization, and these catalysts are poisoned by acetylene. Unfortunately, typical ethylene feedstocks in petrochemical production are contaminated by a few per cent of acetylene, and so a 103 means is necessary to remove this contamination prior to the polymerization step. This is typically 104 105 accomplished by hydrogenation with a metal catalyst at temperatures of 400K and pressures of a few bar. 106 The effective removal of hydrogen at low temperatures on Titan therefore requires an efficient and 107 unknown catalyst. If such a catalyst is nonbiological, its discovery would be of both industrial as well as

109 Another possibility, not excluded by present data, is that living processes may perform such catalysis. Indeed, kinetic inhibition of exothermic reactions is an ideal setting for life – albeit of necessity for Titan, 110 111 not the water-mediated 'Life as we know it' (LAWKI). In a seminal paper that considered the range of possible chemistries for life, Benner et al. (2004) - see also Bains (2004) - speculated that hydrocarbons 112 113 that are naturally liquid on Titan could be a solvent for life. Benner et al. (2004) noted that the organic 114 reactivity in hydrocarbon solvents is no less versatile than in water, and indeed the ability to exclude 115 water is an important aspect of many catalytic sites. Schulze-Makuch and Grinspoon (2005) and McKay 116 and Smith (2005) noted that the photochemically produced organics in Titan's atmosphere would produce energy if reacted with atmospheric H₂, and that this could be a source of biological energy. McKay and 117 118 Smith (2005) quantified the energy released from such reactions. Hydrogenation of C_2H_2 provided a particularly energetic reaction, with 334 kJ per mole of C_2H_2 consumed. This can be compared to the 119

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planetological interest.

120 minimum energy required to power methanogen growth on Earth of ~ 40 kJ mole⁻¹, determined by Kral et

121 al. (1998), or the energy from the reaction of O_2 with CH₄, which produces ~900 kJ mole⁻¹.

122 Photosynthesis has even been considered in hydrogen-rich atmospheres (Bains et al., 2014).

123 In addition to H₂, McKay and Smith (2005) suggested that methane-based life on Titan would also

124 consume acetylene and ethane. There seems to be evidence for depletion of acetylene and ethane on

125 Titan. The data that suggest that there is less ethane on Titan than expected is well established. (Lorenz et

al. 2008). Photochemical models have predicted that Titan should have a layer of ethane sufficient to

127 cover the entire surface to a thickness of many meters but Cassini has found no such layer. Clark et al.

(2010) find a lack of acetylene on the surface despite its expected production in the atmosphere and
subsequent deposition on the ground. There was also no evidence of acetylene in the gases released from
the surface after the Huygens Probe landed (Niemann et al. 2005, Lorenz et al. 2006). Thus, the evidence

131 for less ethane and less acetylene that expected seems clear.

Cornet et al. (2015) review photochemical models for Titan and list production rates for C_2H_2 , from 0.32 to 1.2 x 10⁹ cm⁻² s⁻¹ (e.g., Toublanc et al. 1995, Krasnopolsky 2009) and C_2H_6 , from 1.2 to 15 x 10⁹ cm⁻² s⁻¹ 134¹; thus if ~20% of the available C_2H_2 and C_2H_6 is consumed by methanogens, the corresponding H_2 consumption (10⁹ cm⁻² s⁻¹) should significantly deplete the H_2 profile in the lowest few kilometers of the atmosphere (e.g. McKay, 2016).

There is of course a rich history of considering atmospheric composition as an indicator of life (e.g.
Lovelock, 1965). Whether the processes controlling the vertical profile and horizontal distribution of
hydrogen on Titan are biological or only physico-chemical, the interest in exploring this compound's
distribution is evident. We now consider how hydrogen variations might be determined in future
missions.

143 Measurement Techniques - Review

A future orbiter to Titan (e.g. that studied in the 2007 Titan Flagship study, Leary et al., 2007) could carry an infrared spectrometer able to measure hydrogen abundance in the troposphere in the same way as Cassini and Voyager. An orbiter (at an altitude of e.g. 1500km) would be able to make measurements with better horizontal resolution than typical during Cassini encounters, and of course could obtain much more complete coverage in time and space. However, such remote measurements have very limited ability to discriminate vertical variations, since the weighting functions of the N₂-H₂ dimer transitions are rather broad (e.g. Courtin et al., 2012 figure 4), spanning ~half the troposphere.

151 In-situ missions, such as landers, aircraft and balloons have considerable appeal at Titan (e.g. Lorenz,

152 2000) and could provide information on much smaller scales than orbiter data. It is of course possible

also to measure the hydrogen abundance on Titan with a mass spectrometer, as done on Huygens. A

154 general challenge with mass spectrometry to measure hydrogen abundance is that hydrogen is obviously a

155 fragmentation product of many hydrocarbons (notably methane) requiring instrument corrections (e.g.

156 Niemann et al., 2010). A further point, related to at least some instrument designs, is that there may exist

157 a limited dynamic range (e.g. a factor of 100) in masses that can be practicably analyzed. Hydrogen is the

158 only molecular species (M=2) of interest below the mass of methane (M=16), yet there are likely

159 hundreds or thousands of potential compounds of interest on Titan with masses in the hundreds or

160 thousands. Thus designing a mass spectrometer to handle hydrogen may mean compromising its ability to

161 study high-molecular-weight compounds. A mass spectrometer in a dense atmosphere like Titan's also

162 requires pumping to maintain an internal vacuum, and thus it is a somewhat elaborate and resource-

163 hungry instrument for making 'routine' monitoring measurements of hydrogen deep in the atmosphre.

164 The natural choice to measure hydrogen abundance in-situ on Titan is with a hydrogen sensor. Solid-state 165 sensors that respond to the abundance of a single gas (or a limited range of species) have been used since 166 the earliest days of solar system exploration, e.g. on the Venera probes. More recently, an amperometric

167 oxygen sensor was flown on the Thermal and Evolved Gas Analyzer (TEGA) on Mars Polar Lander (e.g.
168 Boynton et al., 2001) and a capacitive water vapor (i.e. humidity) probe was flown on the Phoenix Mars
169 lander (e.g. Zent et al., 2009).

170 A wide range of hydrogen sensor types exist (e.g. see the review by Hubert et al., 2011). While some 171 familiar types (e.g. the "pellistor", which relies on the catalytic combustion of gas in an oxygen-bearing 172 atmosphere, or thermal conductivity based devices which are insufficiently sensitive to the small [sub-1%] concentrations on Titan) are unsuitable in this application, significant progress in the last couple of 173 decades has been made in Palladium- and Platinum-based sensors. In these, hydrogen diffuses into and is 174 accommodated in the metal lattice (effectively, hydrogen is 'soluble' in the metal - in fact, the 175 176 accommodation is a two-step process, with dissociation on the palladium into hydrogen atoms being the 177 rate-limiting step : the atomic hydrogen then diffuses into the palladium lattice, causing an expansion.) The lattice expansion manifests as changes in resistivity or work function which can be rather easily 178 179 sensed by building resistors or Shottky diodes around the relevant metals. Because the absorption of hydrogen into and out of the metal is a purely physical process, it is reversible and does not rely on an 180 oxidizing atmosphere. Indeed, such hydrogen sensors have been used to measure the purging of rocket 181 182 propellant feed lines by nitrogen or helium (e.g. Hunter et al., 1998), and in equipment on the Space Station (M'Sadoques and Makel, 2005). In the following section, we describe tests of an example sensor 183 184 suitable for Titan application.

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186 Hydrogen Sensor Tests

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Before contemplating use of hydrogen sensor at Titan, it is important to establish the conditions underwhich it may operate. It is further important to understand the specificity of the sensor, to determine in

190 particular that the sensor does not respond significantly to variations in methane. To that end, we have 191 conducted tests at NASA's Ames Research Center of a palladium sensor in a Titan simulation chamber. 192 These Titan simulation facilities have been used in prior studies of Titan photochemistry, and slightly 193 modified to test the H₂ sensor either under flow conditions using mass flow controllers or under an 194 isolated static condition of a gas mixture. The whole chamber can be cooled down to 77 K with 195 immersing into liquid nitrogen.

196 The sensor incorporates both resistive and Shottky diode elements to span a wide range of hydrogen abundance measurements (from a few ppm to 100%), but with the relatively large abundances in the 197 present work, only the resistor output is reported here. The sensor is also equipped with a resistance 198 temperature detector on the die itself, to permit precise closed-loop thermal control. Since the hydrogen 199 200 sensor signal is a strong function of the temperature, this temperature regulation is essential. The sensor die is mounted on a PEEK (Polyether ether ketone) standoff to minimise heat leak and thus heater power, 201 202 and is installed in a TO-5 can (a ~9mm diameter metal housing) with an aperture to allow gas exchange 203 (figure 2).

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Figure 2. The sensor fabricated by Makel Engineering, Inc. for this test : the die can be seen at left on the
PEEK standoff. The TO-5 housing at right is ~9mm in diameter.

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Since the transducer response is based on the absorption of elemental hydrogen into the metal, it responds to the partial pressure of hydrogen, rather than the mixing ratio per se. Since total pressure is easy to measure (and varies little on Titan, for a given altitude), the partial pressure inferred from the sensor reading is readily converted into a mixing ratio. The sensor was installed in the Titan chamber (a cylindrical vessel which could be chilled) and gas mixtures delivered by a flow controller. Four different gas mixtures with various H₂ contents in nitrogen were used. Step changes in gas mixture result in corresponding changes in the sensor output (figure 3).



Figure 3. Sensor response (a) to step changes in hydrogen concentration (sensor at 50°C) up to a few
per cent (b). In this instance the sensor reading is in counts from a 12-bit analog-to-digital converter
reading a voltage formed by the Palladium resistor in a potential divider circuit, with higher counts
corresponding to higher resistance and hydrogen abundance. The brief negative-going transients are test

artifacts associated with the adjustment of the nitrogen-hydrogen flow controller to maintain a constantpressure condition. The response time is approximately 40s.

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As expected, the response time of the sensor is strongly temperature-dependent. The physical limitation on the process is the finite size of the transducer material, the dissociation of H₂ molecules on the surface , and the diffusion coefficient of hydrogen in the material. Manufacture as a thinner structure might improve matters, as can higher temperature operation. Measurements performed in pure H₂ flow revealed response times of ~ 40 secs at 50 °C (323 K) ~ 80 sec at 22.5 °C (295 K), and some ~250 sec at 5 °C (278 K). Clearly, operation at Titan surface temperature would be impractical, although the sensor is not adversely affected by exposure to such conditions.

234 The quantitative output of the device, like many solid state gas sensors, is roughly logarithmic (see figure 4). It is evident in figure 3 that the sensor responds well to 1-3% concentrations of hydrogen at 1 bar 235 (thus an order of magnitude higher partial pressure than expected at Titan's surface, ~ 1500 ubar, or 150 236 Pa). Figure 4 demonstrates that the output variation is quite measureable with 1000 ppm abundance even 237 at 10 mbar (1 kPa) total pressure, thus 100x lower partial pressure than Titan's surface (10 mbar 238 239 corresponds to ~100km altitude on Titan). Thus this type of sensor would be useful for operation in at least the lower stratosphere of Titan during parachute descent, as well as for surface and tropospheric 240 241 measurements.



Figure 4. Output as a function of (total) pressure for a simulated Titan atmosphere (97.4% N₂, 2.7% CH₄,
1180 ppm H₂). The gas was at 77K but the sensor itself was heated to 323 K. As expected, the output is
roughly logarithmic in partial pressure.

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The heater power required to maintain the sensor at 323 K at 1 bar of the Titan gas mixture was 210 mW
when the gas was at 77K. Reducing the operating temperature to 278 K (with a 6x penalty in response
time, as indicated above) lowers the heater power requirement only to 160 mW, a rather modest saving.
Thus operation at 323 K or higher is recommended. For comparison for terrestrial applications, with the

gas at 278 K, sensor operation at 323 K requires only 34 mW. In steady flow conditions, the heater
maintained the sensor to within about 0.1K of the set point.

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254 Accommodation on a vehicle at Titan should shield the sensor from draughts, to prevent wind (typically 255 0-0.5 m/s at Titan's surface, e.g. Folkner et al., 2006; Lorenz et al., 2012) from causing fluctuations in the sensor temperature via forced convective cooling. A simple wind shield a couple of cm across (perhaps a 256 wire mesh screen) will suffice, it being straightforward to permit an exchange time of air rapid enough not 257 to slow the sensor response (~50s) without requiring large internal flowspeeds. Note that in the absence 258 259 of wind, at least, heater power for a given pressure/temperature condition on Titan should be slightly 260 lower than at Earth, since Titan's lower gravity will make free convective heat transfer slightly less 261 effective (e.g. Lorenz, 2016).

Finally, we verified that the sensor (as would be expected from the sensing modality, which is specific to 262 the small size of the hydrogen atom) had negligible sensitivity to methane variations. Note that there can 263 be in principle (Soundarrajan and Schweighardt, 2009) some sensitivity to acetylene if the sensor is hot 264 enough to cause dissociation at its surface, but the concentration of gaseous acetylene in Titan's lower 265 266 atmosphere is only a few parts per million (e.g. Hörst, 2017). It should be noted that for this Titan 267 application, the relative response of the sensor to hydrogen variations is in any case more important than 268 the absolute value, in that the purpose as described in the next section is to look for gradients in 269 A test was run wherein about 40 standard cc of nitrogen per minute (sccm) was flowed past abundance. the sensor (here at 298K) while the hydrogen flow was flipped between 1 and 2 sccm. The 'square wave' 270 271 response was essentially unaffected when 2, 4 or 6 sccm of CH₄ flow was added.

As figure 5 shows, the sensor output barely changed in response to CH_4 injections, while the H_2 response was strong. Sequential hydrogen on-off readings were consistent [2016.5, 2015.8, 2013.3] and [2058.9,

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274 2055.2, 2056.2] respectively, with standard deviations within each sample of ~3 points. The change in 275 output driven by the large shift in CH_4 abundance is, then only ~3 points at most, comparable with the 276 sample deviation of individual readings. This may be compared with the consistent variation in output of 277 ~40 points associated with the hydrogen abundance variation (of a factor of ~2) – the sensitivity is at least 278 an order of magnitude less for methane than hydrogen. Note that, as for other effects like pressure, local 279 measurements of the methane abundance could be used to correct the sensor output in any case, although 280 we have only been able to set an upper bound on this cross-sensitivity.



Figure 5. Sensor response to H₂ abundance, and lack of response to CH₄. Apart from a brief transient at
5550s (again, a chamber pressure artifact caused by the adjustment of the flow controller) the response to

methane is barely detectable. The red sections denote where averages and standard deviations werecalculated (see text).

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288 In-Situ Observational Strategies at Titan

The instrument at hand is a point sensor, measuring the hydrogen abundance at a single location and time. In order to determine the question of interest, whether there are sources or sinks of hydrogen in the surface, the sensor can be used in a number of ways. These modalities parallel the exploration of terrestrial gas exchanges (such as volcanic gases, or methane, radon etc.) or possible methane sources on Mars.

First, following the initial speculation of Mckay and Smith (2007), the vertical gradient in hydrogen abundance can be measured by transporting the sensor vertically. This occurs naturally during the descent of a lander or a probe, often under a parachute, but could also be accomplished repeatedly by a Montgolfiere (hot air balloon) using the 'waste' heat from a radioisotope power source and with a controlled vent to modulate its buoyancy. Heavier-than-air flight is also possible, as in the AVIATR fixed-wing concept (Barnes et al., 2012: like all long-lived in-situ exploration concepts at Titan, this would also use radioisotope power, in this instance a proposed Stirling generator).

The proposed Dragonfly relocatable rotorcraft lander (Lorenz et al., 2018; Turtle et al., 2018), currently undergoing a Phase A study for possible implementation in NASA's New Frontiers Program, is another platform that could perform hydrogen profiling. This vehicle could make repeated ascents and descents through the planetary boundary layer and thereby constrain the magnitude of surface sources or sinks of hydrogen from the same type of models explored by Strobel et al. (2010) and McKay (2016).

The relocation capability of Dragonfly allows long-term measurements on the surface at different locations. The combination of wind speed and direction knowledge from DraGMet (the Dragonfly Geophysics and Meteorology package) with hydrogen abundance would allow the identification of possible hydrogen plume sources. For example, if a feature such as an impact crater or cryovolcano were considered a possible plume source, then enhanced hydrogen abundance when the lander was downwind of the source might support that hypothesis. Relocating the lander in combination with wind direction variations would allow the structure of the plume to be understood.

313 A final method, used in many studies of surface-atmosphere exchange on Earth, is the eddy-covariance technique e.g. Fairall (2000). Specifically, if a vertical gradient in a mixing ratio exists, then turbulent 314 315 eddies will advect parcels of air from lower and higher elevations past the sensor, whose output will 316 therefore have short-term fluctuations. The amplitude of these fluctuations relates to the surface flux in ways that can be modelled and/or measured (specifically, a cross-correlated high resolution time series of 317 vertical wind velocity and the hydrogen abundance would yield the flux directly : in practice vertical 318 winds are often estimated from horizontal winds and there are limitations on the method from the finite 319 height and finite time response of the sensor). Since the preferred sample rate for eddy flux 320 321 measurements is $\sim 10 \text{ u/z}$, where u is the horizontal windspeed and z the measurement height, a landed instrument should be sampled at ~10Hz, so the sensor tested here is too slow. These limitations aside, it 322 323 is obvious that if there are no horizontal or vertical gradients in hydrogen abundance, the output of a fixed 324 sensor should not exhibit fluctuations, so even low-resolution time series data provide a simple qualitative 325 test of the possible existence of sources or sinks which can be quantified with the aid of models (e.g. as done for water vapor in the Martian surface, Savijärvi et al., 2015) 326

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329 Conclusions

- 330 Our analysis of the Cassini data and theoretical models indicate that hydrogen in the lower atmosphere of Titan could be variable due atmospheric transport mechanisms, losses due to surface reactions, and 331 332 subsurface sources and sinks. We conclude that detection and quantification of these hydrogen variations 333 over spatial and temporal scales could point to new atmospheric processes, catalytic surface chemistries, 334 and possibly biological consumption. We have identified Palladium-based hydrogen sensors as capable of making these measurement in situ. In laboratory tests with Titan-like gas mixtures we have shown 335 specificity to hydrogen independent of nitrogen and methane and a rapid response time if heated to 50°C. 336 We conclude that these low mass sensors could provide a proven and robust method for in-situ 337 determination of variation of hydrogen in Titan's lower atmosphere on future missions. 338 339 Acknowledgements 340 We thank Brandon Thacker for assistance with the sensor installation at Ames, and Peter Bedini, Scott 341 342 Murchie and Walt Bender at APL for project coordination. Work at APL was supported by internal funds in the context of the Dragonfly New Frontiers Phase A study award from NASA Headquarters. 343 344 345 References 346 347
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Reviews measurement strategies on a mobile lander